



# SNS COLLEGE OF TECHNOLOGY

(An Autonomous Institution)

Approved by AICTE, New Delhi, Affiliated to Anna University, Chennai

Accredited by NAAC-UGC with 'A++' Grade (Cycle III) &

Accredited by NBA (B.E - CSE, EEE, ECE, Mech & B.Tech.IT)

COIMBATORE-641 035, TAMIL NADU



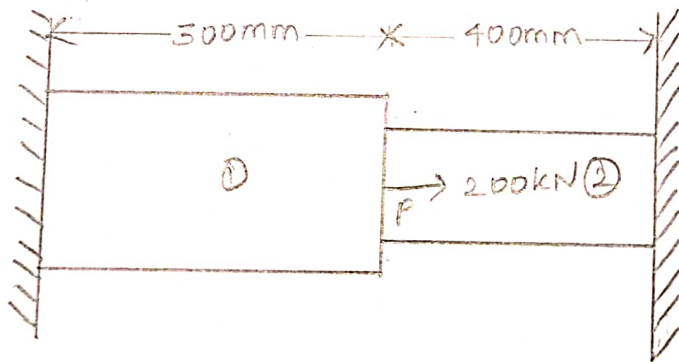
## DEPARTMENT OF AEROSPACE ENGINEERING

Faculty Name : Dr.M.Subramanian,  
Prof & Head/ Aerospace Academic Year : 2024-2025 (Odd)  
Year & Branch : III Aerospace Semester : V  
Course : 19ASB302 - Finite Element Method for Aerospace

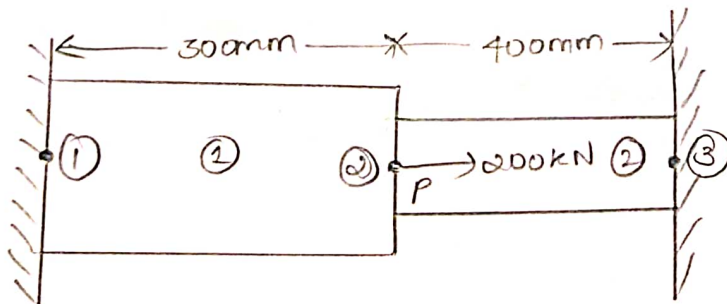
① Consider a bar as shown in fig.(i). An axial load of 200kN is applied at point p. Take  $A_1 = 2400 \text{ mm}^2$ ,  $E_1 = 70 \times 10^9 \text{ N/m}^2$ ,  $A_2 = 600 \text{ mm}^2$ ,  $E_2 = 200 \times 10^9 \text{ N/m}^2$ .

Calculate the following:

- The nodal displacement at point p.
- Stress in each material
- Reaction force.



GIVEN:



Prepared: Dr. M. Subramanian/Professor & Head Aerospace Engineering



# SNS COLLEGE OF TECHNOLOGY

(An Autonomous Institution)

Approved by AICTE, New Delhi, Affiliated to Anna University, Chennai

Accredited by NAAC-UGC with 'A++' Grade (Cycle III) &

Accredited by NBA (B.E - CSE, EEE, ECE, Mech & B.Tech.II)

COIMBATORE-641 035, TAMIL NADU



## DEPARTMENT OF AEROSPACE ENGINEERING

Faculty Name : Dr.M.Subramanian, Academic Year : 2024-2025 (Odd)  
Year & Branch : Prof & Head/ Aerospace Semester : V  
Course : III Aerospace  
: 19ASB302 - Finite Element Method for Aerospace

Area of element (1),  $A_1 = 2400\text{mm}^2$

Area of element (2),  $A_2 = 600\text{mm}^2$

Length of element (1),  $l_1 = 300\text{mm}$

Length of element (2),  $l_2 = 400\text{mm}$

Young's modulus of element (1),  $E_1 = 70 \times 10^9 \text{N/m}^2$   
 $= 70 \times 10^3 \text{N/mm}^2$

Young's modulus of element (2),  $E_2 = 200 \times 10^9 \text{N/m}^2$   
 $= 200 \times 10^3 \text{N/mm}^2$

Point load,  $p = 200\text{kN} = 200 \times 10^3 \text{N}$

TO FIND:

- (i) Nodal displacement at point, p i.e.,  $u_2$ .
- (ii) Stress in each material,  $\sigma_1$  and  $\sigma_2$
- (iii) Reaction forces,  $R_1, R_2$ .

SOLUTION:

Prepared: Dr. M. Subramanian/Professor & Head Aerospace Engineering



# SNS COLLEGE OF TECHNOLOGY

(An Autonomous Institution)

Approved by AICTE, New Delhi, Affiliated to Anna University, Chennai

Accredited by NAAC-UGC with 'A++' Grade (Cycle III) &

Accredited by NBA (B.E - CSE, EEE, ECE, Mech & B.Tech.IT)

COIMBATORE-641 035, TAMIL NADU



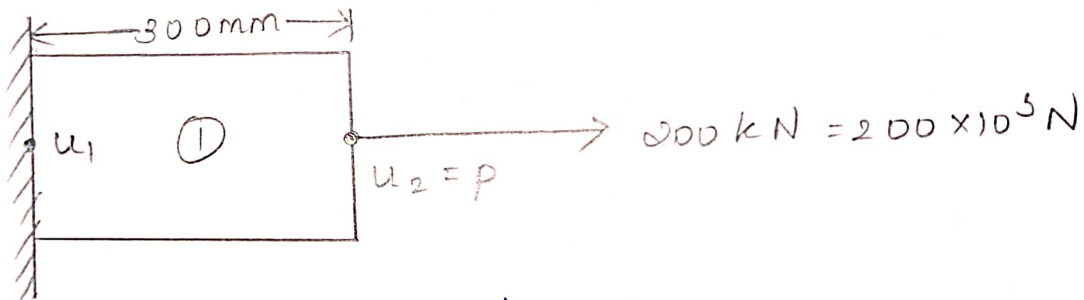
## DEPARTMENT OF AEROSPACE ENGINEERING

Faculty Name : **Dr.M.Subramanian,** Academic Year : **2024-2025 (Odd)**  
 Year & Branch : **III Aerospace** Semester : **V**  
 Course : **19ASB302 – Finite Element Method for Aerospace**

Finite element equation for one dimensional two noded bar element is given by,

$$\begin{Bmatrix} F_1 \\ F_2 \end{Bmatrix} = \frac{AE}{l} \begin{bmatrix} 1 & -1 \\ -1 & 1 \end{bmatrix} \begin{Bmatrix} u_1 \\ u_2 \end{Bmatrix}$$

For example element 1:



Finite element equation is,

$$\frac{A_1 E_1}{l_1} \begin{bmatrix} 1 & -1 \\ -1 & 1 \end{bmatrix} \begin{Bmatrix} u_1 \\ u_2 \end{Bmatrix} = \begin{Bmatrix} F_1 \\ F_2 \end{Bmatrix}$$

$$\Rightarrow \frac{2400 \times 10 \times 10^3}{300} \begin{bmatrix} 1 & -1 \\ -1 & 1 \end{bmatrix} \begin{Bmatrix} u_1 \\ u_2 \end{Bmatrix} = \begin{Bmatrix} F_1 \\ F_2 \end{Bmatrix}$$

$$\Rightarrow 1 \times 10^5 \begin{bmatrix} a_{11} & a_{12} \\ a_{21} & a_{22} \end{bmatrix} \begin{Bmatrix} u_1 \\ u_2 \end{Bmatrix} = \begin{Bmatrix} F_1 \\ F_2 \end{Bmatrix} \dots \dots \textcircled{1}$$

Prepared: Dr. M. Subramanian/Professor & Head Aerospace Engineering



# SNS COLLEGE OF TECHNOLOGY

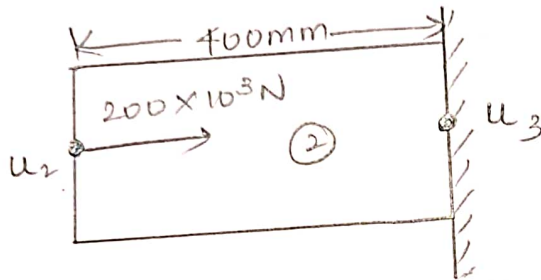
(An Autonomous Institution)  
Approved by AICTE, New Delhi, Affiliated to Anna University, Chennai  
Accredited by NAAC-UGC with 'A++' Grade (Cycle III) &  
Accredited by NBA (B.E - CSE, EEE, ECE, Mech & B.Tech.IT)  
COIMBATORE-641 035, TAMIL NADU



## DEPARTMENT OF AEROSPACE ENGINEERING

Faculty Name : Dr.M.Subramanian, Academic Year : 2024-2025 (Odd)  
Year & Branch : Prof & Head/ Aerospace Semester : V  
Course : 19ASB302 – Finite Element Method for Aerospace

For element 2: Finite element equation is,



$$\frac{A_2 E_2}{l_2} \begin{bmatrix} 1 & -1 \\ -1 & 1 \end{bmatrix} \begin{Bmatrix} u_2 \\ u_3 \end{Bmatrix} = \begin{Bmatrix} F_2 \\ F_3 \end{Bmatrix}$$

$$\Rightarrow \frac{600 \times 200 \times 10^3}{400} \begin{bmatrix} 1 & -1 \\ -1 & 1 \end{bmatrix} \begin{Bmatrix} u_2 \\ u_3 \end{Bmatrix} = \begin{Bmatrix} F_2 \\ F_3 \end{Bmatrix} \text{----- (2)}$$

Assemble the finite elements. i.e., assemble the finite element equations (1) and (2).

$$\Rightarrow 1 \times 10^5 \begin{bmatrix} a_{11} & a_{12} & a_{13} \\ 5.6 & -5.6 & 0 \\ a_{21} & a_{22} & a_{23} \\ -5.6 & 5.6+3 & -3 \\ a_{31} & a_{32} & a_{33} \\ 0 & -3 & 3 \end{bmatrix} \begin{Bmatrix} u_1 \\ u_2 \\ u_3 \end{Bmatrix} = \begin{Bmatrix} F_1 \\ F_2 \\ F_3 \end{Bmatrix}$$

Prepared: Dr. M. Subramanian/Professor & Head Aerospace Engineering



# SNS COLLEGE OF TECHNOLOGY

(An Autonomous Institution)

Approved by AICTE, New Delhi, Affiliated to Anna University, Chennai

Accredited by NAAC-UGC with 'A++' Grade (Cycle III) &

Accredited by NBA (B.E - CSE, EEE, ECE, Mech & B.Tech.IT)

COIMBATORE-641 035, TAMIL NADU



## DEPARTMENT OF AEROSPACE ENGINEERING

Faculty Name : Dr.M.Subramanian, Academic Year : 2024-2025 (Odd)  
Prof & Head/ Aerospace  
Year & Branch : III Aerospace Semester : V  
Course : 19ASB302 – Finite Element Method for Aerospace

$$\Rightarrow 1 \times 10^5 \begin{bmatrix} 5.6 & -5.6 & 0 \\ -5.6 & 8.6 & -3 \\ 0 & -3 & 3 \end{bmatrix} \begin{Bmatrix} u_1 \\ u_2 \\ u_3 \end{Bmatrix} = \begin{Bmatrix} F_1 \\ F_2 \\ F_3 \end{Bmatrix} \quad \text{--- (3)}$$

↓  
[K]

NOTE: The bar has 3 nodes. Each node has single degree of freedom. So, the global stiffness matrix [K] size is 3x3, The properties of the stiffness matrix are also satisfied.

(i) [K] matrix is symmetric.

(ii) The sum of elements in any column is equal to zero.

Applying boundary conditions :

Displacements at node 1 and node 3 are zero. So,  $u_1 = u_3 = 0$ . A Load of  $200 \times 10^3 \text{ N}$  is acting at node 2. So,  $F_2 = 200 \times 10^3 \text{ N}$ . Self-weight is neglected .i.e.,  $F_1 = F_3 = 0$ . Substitute

Prepared: Dr. M. Subramanian/Professor & Head Aerospace Engineering



# SNS COLLEGE OF TECHNOLOGY

(An Autonomous Institution)

Approved by AICTE, New Delhi, Affiliated to Anna University, Chennai

Accredited by NAAC-UGC with 'A++' Grade (Cycle III) &

Accredited by NBA (B.E - CSE, EEE, ECE, Mech & B.Tech.IT)

COIMBATORE-641 035, TAMIL NADU



## DEPARTMENT OF AEROSPACE ENGINEERING

Faculty Name : Dr.M.Subramanian, Academic Year : 2024-2025 (Odd)  
 Prof & Head/ Aerospace  
 Year & Branch : III Aerospace Semester : V  
 Course : 19ASB302 - Finite Element Method for Aerospace

$u_1, u_3$  and  $F_1, F_2$  and  $F_3$  values in equation (3)

$$(3) \Rightarrow 1 \times 10^5 \begin{bmatrix} 5.6 & -5.6 & 0 \\ -5.8 & 8.6 & -3 \\ 0 & -3 & 3 \end{bmatrix} \begin{Bmatrix} 0 \\ u_2 \\ 0 \end{Bmatrix} = \begin{Bmatrix} 0 \\ 2 \times 10^5 \\ 0 \end{Bmatrix}$$

In the above equation,  $u_1 = 0$ . So, neglect first row and first column of  $[k]$  matrix.  
 $u_3 = 0$ , so, neglect third row and third column of  $[k]$  matrix. The final reduced equation

is,

$$1 \times 10^5 [8.6] \{u_2\} = \{2 \times 10^5\}$$

$$\Rightarrow 8.6 \times 10^5 u_2 = 2 \times 10^5$$

$$8.6 u_2 = 2$$

$$u_2 = 0.2325 \text{ mm}$$

Stress in each element:

We know that, stress,  $\sigma = E \frac{du}{dx}$

For element (1), stress,  $\sigma_1 = E_1 \times \frac{u_2 - u_1}{300} = 70 \times 10^3 \times \frac{(0.2325 - 0)}{300}$

Prepared: Dr. M. Subramanian/Professor & Head Aerospace Engineering



# SNS COLLEGE OF TECHNOLOGY

(An Autonomous Institution)

Approved by AICTE, New Delhi, Affiliated to Anna University, Chennai

Accredited by NAAC-UGC with 'A++' Grade (Cycle III) &

Accredited by NBA (B.E - CSE, EEE, ECE, Mech & B.Tech.IT)

COIMBATORE-641 035, TAMIL NADU



## DEPARTMENT OF AEROSPACE ENGINEERING

Faculty Name : Dr.M.Subramanian, Academic Year : 2024-2025 (Odd)  
 Year & Branch : Prof & Head/ Aerospace Semester : V  
 Course : III Aerospace  
 : 19ASB302 – Finite Element Method for Aerospace

$$\sigma_1 = 54.25 \text{ N/mm}^2$$

For element(2), Stress,  $\sigma_2 = E_2 \times \frac{u_3 - u_2}{l_2}$

$$= 200 \times 10^3 \times \frac{(0 - 0.2325)}{400}$$

$$\Rightarrow \sigma_2 = -116.25 \text{ N/mm}^2 \left[ \begin{array}{l} \text{compressive} \\ \text{stress is acting} \end{array} \right]$$

Reaction force:

We know that,

$$\text{Reaction force, } \{R\} = [k] \{u^*\} - \{F\}$$

$$\Rightarrow \begin{Bmatrix} R_1 \\ R_2 \\ R_3 \end{Bmatrix} = 1 \times 10^5 \begin{bmatrix} 5.6 & -5.6 & 0 \\ -5.6 & 8.6 & -3 \\ 0 & -3 & 3 \end{bmatrix} \begin{Bmatrix} u_1 \\ u_2 \\ u_3 \end{Bmatrix} - \begin{Bmatrix} F_1 \\ F_2 \\ F_3 \end{Bmatrix}$$

$$\Rightarrow \begin{Bmatrix} R_1 \\ R_2 \\ R_3 \end{Bmatrix} = 1 \times 10^5 \begin{bmatrix} 5.6 & -5.6 & 0 \\ -5.6 & 8.6 & -3 \\ 0 & -3 & 3 \end{bmatrix} \begin{Bmatrix} 0 \\ 0.2325 \\ 0 \end{Bmatrix} - \begin{Bmatrix} 0 \\ 2 \times 10^5 \\ 0 \end{Bmatrix}$$

Prepared: Dr. M. Subramanian/Professor & Head Aerospace Engineering



# SNS COLLEGE OF TECHNOLOGY

(An Autonomous Institution)

Approved by AICTE, New Delhi, Affiliated to Anna University, Chennai

Accredited by NAAC-UGC with 'A++' Grade (Cycle III) &

Accredited by NBA (B.E - CSE, EEE, ECE, Mech & B.Tech.IT)

COIMBATORE-641 035, TAMIL NADU



## DEPARTMENT OF AEROSPACE ENGINEERING

Faculty Name : Dr.M.Subramanian, Academic Year : 2024-2025 (Odd)  
Prof & Head/ Aerospace  
Year & Branch : III Aerospace Semester : V  
Course : 19ASB302 – Finite Element Method for Aerospace

$$= 1 \times 10^5 \begin{bmatrix} -1.302 \\ 2 \\ -0.6975 \end{bmatrix} - \begin{bmatrix} 0 \\ 2 \times 10^5 \\ 0 \end{bmatrix} = \begin{bmatrix} -1.302 \times 10^5 \\ 2 \times 10^5 \\ -0.6975 \times 10^5 \end{bmatrix} - \begin{bmatrix} 0 \\ 2 \times 10^5 \\ 0 \end{bmatrix}$$

$$\begin{Bmatrix} R_1 \\ R_2 \\ R_3 \end{Bmatrix} = \begin{Bmatrix} -1.302 \times 10^5 \\ 0 \\ -0.6975 \times 10^5 \end{Bmatrix}$$

$$\Rightarrow R_1 = -1.302 \times 10^5 \text{ N}$$

$$R_2 = 0 \text{ N}$$

$$R_3 = -0.6975 \times 10^5 \text{ N}$$

We know that, Reaction force is equivalent and opposite to the applied force.

Verification:

$$R_1 + R_2 + R_3 = -1.302 \times 10^5 + 0 - 0.6975 \times 10^5$$

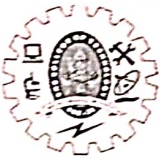
$$= -2.00 \times 10^5 \text{ N (Applied force)}$$

Result: (i) Nodal displacement at point, p,

$$\text{i.e., } u_2 = 0.2325 \text{ mm.}$$

Prepared: Dr. M. Subramanian/Professor & Head Aerospace Engineering





## DEPARTMENT OF AEROSPACE ENGINEERING

Faculty Name : Dr.M.Subramanian, Academic Year : 2024-2025 (Odd)  
Prof & Head/ Aerospace  
Year & Branch : III Aerospace Semester : V  
Course : 19ASB302 – Finite Element Method for Aerospace

(ii) Stress in each material,

$$\sigma_1 = 54.25 \text{ N/mm}^2 \text{ (tensile)}$$

$$\sigma_2 = -116.25 \text{ N/mm}^2 \text{ (compressive)}$$

(iii) Reaction forces,

$$R_1 = -1.302 \times 10^5 \text{ N}; R_2 = 0$$

$$R_3 = -0.695 \times 10^5 \text{ N}$$